ASTEROID MASS DETERMINATION WITH THE GAIA MISSION

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Abstract. The ESA astrometric mission Gaia, due for launch in late 2011, will observe a very large number of asteroids (about 300,000 brighter than V = 20), the majority lying in the main-belt and with an unprecedented positional precision (at the sub-milliarcsecond level). Such high precision astrometry will enable to derive direct measures of the masses of the largest asteroids of which their precise determination will be of utmost significance for the knowledge of physical properties of asteroids. The method for computing the masses is based on the analysis of orbital perturbations during close encounters between massive asteroids (perturbers) and several smaller minor planets (targets). From given criteria of close approaches selection, we give the list of asteroids for which the mass can be determined, and the expected precision of these masses at mission completion.

1 Introduction

Gaia is an astrometric cornerstone mission of the European Space Agency. With a launch due in late 2011, Gaia will have a much more ambitious aim than its precursor Hipparcos: obtain a "3D census" of our galaxy with astrometric, photometric and spectroscopic observations. It will pinpoint its sources with an unprecedented positional precision (at the sub-milliarcsecond level for single observation) which will allow it to observe about 300,000 asteroids brighter than V = 20, the majority being in the main-belt. The high precision astrometry will enable us to considerably improve the orbits of almost all observed asteroids, yielding the masses of the largest one from mutual perturbations. The objective of this work is the determination of the masses of a small subset of minor planets, modeling the signature of the gravitational perturbation affecting smaller objects during close encounters. Not only does the method consider multiple close encounters with different targets, but it also simultaneously treats all perturbers together with their target asteroids. Here we explain the impact of these masses on the dynamic and physic of the solar system objects. We give a list of asteroids for which the mass can be determined, and the expected precision of these masses at mission completion from realistic simulations of the Gaia observations (geometry, time sequence, magnitude, ...).

2 On the importance of masses

From the physical point of view, knowing the mass and dimensions (from IRAS or also from Gaia) of an asteroid allows us to determine its bulk density, which includes the space occupied by its pores. The calculation of the porosity, which enables us to estimate the part of void which makes up the asteroid, is of a great interest. It directly depends on the ratio between the bulk density of the asteroid and the grain density of a meteorite possessing a compositional analog. Knowing that during a collision, the internal structure of an asteroid acts upon the stress-wave propagation, which in turn, modifies its constitution, the porosity becomes a precious indicator of the collisional history of asteroids and their internal structure (Britt et al. 2002).

In addition, such measure of the bulk density will shed light on the relationship, if any, between the spectroscopic taxonomic class and the density, or possibly the porosity. Such progress could provide information on their origin and on the formation process of the solar system. If a well defined relationship can be evidenced, we could make reliable density estimates for objects of a known taxonomic characterization.

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In the same way, the comparison between these densities for objects belonging to different taxonomic classes and the density of meteorites may strengthen the idea about the genetic relationship between different classes of meteorites and their supposed asteroid source (Zappala et al. 2002).

From the dynamical point of view, the knowledge of the asteroid masses could allow us to obtain precise long-term orbit of several smaller objects that large asteroids come to perturb. More important, it will enable to increase the accuracy of the solar system ephemeris, knowing that at the present time the limiting factor is the masses of the largest asteroids and main belt in a whole which perturb the planets, and in particular Mars (Standish & Fienga 2002).

3 The mass determination method

The effect of a close encounter as well as is to modify the trajectory of the perturbed bodies or equivalently to change the initial conditions of its motion at some reference time. The observed minus calculated positions (**O-C**) for each observation of minor planets, which is expressed in Gaia longitude λ projected over a given Great Circle, can be linearized and solved by the least squares technique,

$$\mathbf{O} - \mathbf{C} = \mathbf{A} \begin{pmatrix} \delta \mathbf{u_0} \\ \delta \mathbf{m_p} \end{pmatrix} \Rightarrow \begin{pmatrix} \delta \mathbf{u_0} \\ \delta \mathbf{m_p} \end{pmatrix} = (\mathbf{A}^t \mathbf{A})^{-1} \mathbf{A}^t (\mathbf{O} - \mathbf{C})$$
(3.1)

where $\delta \mathbf{u}^0 = (\delta \mathbf{u}_1^0, ..., \delta \mathbf{u}_n^0)^t$ is the correction to initial conditions of the all *n* asteroids with $\delta \mathbf{u}_{\mathbf{k}}^0 = (\delta x^0, \delta y^0, \delta z^0, \delta \dot{x}^0, \delta \dot{y}^0, \delta \dot{z}^0)$ the corrections on the position and velocity of asteroid *k* at the initial time, and $\delta \mathbf{m}_{\mathbf{p}} = (\delta m_1, ..., \delta m_p)^t$ are the corrections on masses of the p perturbers.

The matrix **A** in equation (3.1) represents the partial derivatives of the longitudes λ of the minor planets with respect to their initial parameters,

$$[\mathbf{A}]_{i,j} = \sum_{q=1}^{3} \frac{\partial \lambda_i}{\partial x_q} \frac{\partial x_q}{\partial C_j^0} \quad \text{where} \quad \begin{cases} \mathbf{x}_{q=1,..,3} = (x, y, z) \\ \mathbf{C} = (\mathbf{u}_1^0, .., \mathbf{u}_n^0, m_1, .., m_p) \\ \mathbf{u}_{\mathbf{k}}^0 = (x^0, y^0, z^0, \dot{x}^0, \dot{y}^0, \dot{z}^0) \text{ for the } \mathbf{k}^{\text{th}} \text{ asteroid} \\ \lambda \text{ is the vector made up of every observations} \\ \text{of all asteroids.} \end{cases}$$
(3.2)

The expression $\partial \lambda_i / \partial x_q$ is determined analytically, while the right-hand part of the decomposition $\partial x_q / \partial C_j$ is evaluated numerically. The latter is obtained by expressing the variations of the rectangular heliocentric coordinates $x_{q=1,..,3}$ (in an inertial frame) of each asteroid with respect to the adjustable parameters by integrating the variational equations simultaneously with the equations of motion (Herget 1968).

From equation (3.1), it is possible to give the precision $\sigma(\delta \mathbf{m}_{\mathbf{p}})$ with which we will be able to determine the masses of perturbers $\mathbf{m}_{\mathbf{p}}$ knowing the precision of asteroid positions. If the normal matrix $\mathbf{A}^{t}\mathbf{A}$ is invertible, we obtain a relation between the variance-covariance matrix of the foreseen parameters $\operatorname{cov}(\delta \mathbf{u}^{0}, \delta \mathbf{m}_{\mathbf{p}})^{t}$ and the variance-covariance matrix $\operatorname{cov}(\mathbf{O}-\mathbf{C})$. Each line of the matrix \mathbf{A} corresponding to an asteroid k for an observation i is weighted according to the error on the position $\sigma_{k,i}$ (depending on the magnitude) by a constant σ_{0} and so, the measures of the weighted (**O-C**) have the same variance σ_{0}^{2} . In addition, we make the assumption that the measures on the positions are independent, consequently $\operatorname{cov}(\mathbf{O}-\mathbf{C}) = \sigma_{0}^{2}Id$ and we obtain,

$$\cos\left(\delta \mathbf{u}^{\mathbf{0}}, \delta \mathbf{m}_{\mathbf{p}}\right)^{\iota} = \sigma_0^2 (\mathbf{A}^t \mathbf{A})^{-1}$$
(3.3)

4 The selection of target asteroids

An important step for the mass determination is the selection of suitable target asteroids from which the gravitational signature of their perturbations by a larger asteroid is detectable. We restricted ourselves at

this stage to the first 20,000 numbered asteroids for being potential targets, and considered that a potential perturber has a mass greater than $1 \times 10^{-13} M_{\odot}$ and belongs to the first 2,000 numbered asteroids. A close approach is considered meaningful if the impact parameter b (the minimal distance between the two asteroid trajectories in the case where we do not take into account their mutual perturbations) is smaller than 0.2 AU and the deflection angle θ greater than 5 mas estimated by:

$$\tan\frac{\theta}{2} = \frac{G(m+M)}{v^2b} \tag{4.1}$$

where G is the gravitational constant, M the mass of the perturber, m the mass of the target asteroid and v the relative velocity of the encounter (Fig.1).



Fig. 1. Impulse approximation of a small asteroid perturbed by a larger one in the center-of-mass frame of reference

We selected the target asteroids from a systematic exploration of all the close approaches between perturbers and targets from 2011 to 2017 (see Table 1), so that the whole Gaia mission is covered.

 Table 1. Results of close approaches simulations

 Number of

 perturbers
 149
 perturber and target

 target asteroids
 2508
 simultaneously
 54

5 Results

Calculations have been performed by taking into account realistic simulation of the Gaia observations (geometry, time sequence, magnitude,...). First the number of perturbers with respect to the precision with which we will be able to determine their masses is given in Table 2. The results are expressed in terms of the relative precision $\sigma(m)/m$ where m is the reference mass used in the simulation. Then, we give the ranking of the best relative precisions for the mass determination (Table 3). We can also see the number of targets for each mass determination, the number of close approaches for which one of the two characteristic periods (before or after the closest approach) will be out of the range of Gaia observations for the corresponding target asteroids, the formal precisions $\sigma(m)$, the standard masses m taken for the perturbers and the relative precision.

Table 2. Number of the determined masses with respect to the relative precision

Number of perturbers						
	Total	149				
$\sigma(m)/m < 0.1\%$	2	$\sigma(m)/m < 20\%$	61			
$\sigma(m)/m < 1\%$	7	$\sigma({\rm m})/{\rm m} < 30\%$	75			
$\sigma({\rm m})/{\rm m} < 10\%$	46	$\sigma(m)/m < 50\%$	81			

Aste	roid	number of	close approaches	sigma $\sigma(m)$	mass m	$\sigma(m)/m$
n° IAU	name	target asteroids	not observed	$[{ m M}_{\odot}]$	$[{ m M}_{\odot}]$	
4	Vesta	562	214	4.22×10^{-14}	1.30×10^{-10}	0.033~%
1	Ceres	336	109	1.68×10^{-13}	4.50×10^{-10}	0.037~%
14	Irene	113	31	4.56×10^{-14}	2.60×10^{-11}	0.175~%
27	Euterpe	137	39	2.54×10^{-14}	1.00×10^{-11}	0.254~%
10	Hygiea	211	78	1.32×10^{-13}	4.70×10^{-11}	0.282~%
52	Europa	134	73	9.46×10^{-14}	2.40×10^{-11}	0.394~%
2	Pallas	23	2	5.78×10^{-13}	1.30×10^{-10}	0.445~%
16	Psyche	165	70	1.41×10^{-13}	1.40×10^{-11}	1.010~%
46	Hestia	21	5	1.22×10^{-14}	1.20×10^{-12}	1.020~%
88	Thisbe	21	4	5.41×10^{-14}	5.00×10^{-12}	1.080~%

Table 3. The formal precisions on the 10 best mass determination, ordered in decreasing relative precision

6 Conclusions

This simulation taking into account 149 perturbers and some 2508 target asteroids gives very encouraging results since about 50 masses could be estimated with a precision better than 10%. Keeping in mind that Gaia will actually observe about 300,000 asteroids, so that the number of close approaches is significantly increased, one can expect more mass determinations at that precision level. It is now planned to include these 300,000 asteroids in the simulations as well as more perturbers. Besides, we noticed that an important percentage (>30%) of targets asteroids, for half of the perturbers, was found to be not observed by Gaia, either before or after the close approach time. So, a great part of information from the encounter between the perturber and target asteroids is lost and, a study of what could bring ground-based observations before and after the space mission seems to be essential.

References

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